Our procedure to create

Fire Control Inputs (FCI) and Tabular Fire Tables (TFT)

for Howitzers & Mortars



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FCI & TFT Creation Capabilities and Fire Control Software

Fire Control Input (FCI) Creation

Preparation of input files (FCI) used by the fire control computer (AKB) during computer-aided firing with a weapon system for all kinds of heavy weapons and ammunition duos

- Howitzer
- Mortar
- · Tank

Tabular Firing Table (TFT) Creation

Preparation of Tabular Firing Tables (TFT) for all kinds of heavy weapons and ammunition duos

- Howitzer
- Mortar
- · Tank

Ability to use Fire Control Software

- · NABK-based fire control software
- · EDABK-based fire control software

Firing Test Capabilities

Planning

- · Defining necessary data to be gathered during firing tests
- · Defining necessary instruments and their use
- · Defining the number of rounds and the number of projectiles
- · Defining quadrant elevation, azimuth angle, and propellant temperature ranges

Support & Consultancy

· Participating in firing tests as an observer and adviser

FCI & TFT Creation

Determination of Project Stages

The number of stages and their contents are determined depending on the requirements defined in SOW and the parameters affecting them.

Requirements:

- · Sufficient information about weapon and ammunition
- · Existing FCI files for ammunition and weapon and the conditions for the preparation of these files (if any)
- · Actual fire test results for existing ammunition and weapon (if any)

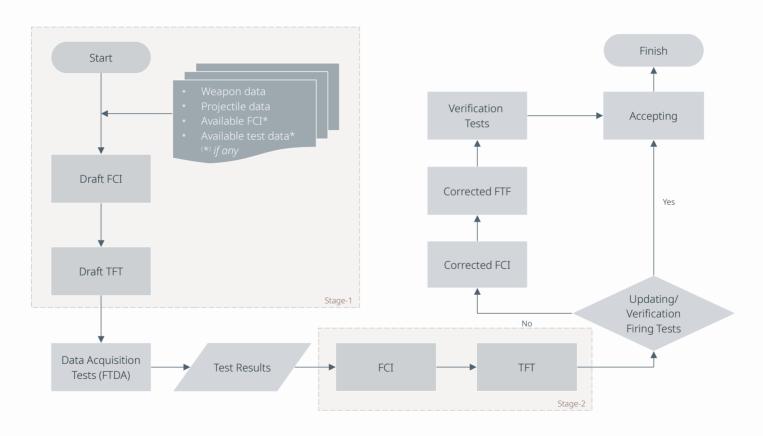
Parameters for Requirements:

- · Difficulty level of engineering problem
- · Technical uncertainties
- · Expected performance and acceptance criteria limitations

Project Stages

	Examining the source documents and input information to decide their adequacy						
Charles 4	Determining the methods to be applied						
Stage-1 Preparation of Draft FCI & TFT	Planning R&D studies if necessary						
	Planning firing test for data acquisition (FTDA) and determining number of necessary number of rounds and shots and conditions						
Firir	ng Tests for Data Acquisition (FTDA)						
Stage-2	Evaluation of results of the FTDA						
Development of FCI & TFT	Updating Drafts and Developing FCI & TFT						
Firing Tests for Verification and Acceptance (FTVA)							

Creation Process

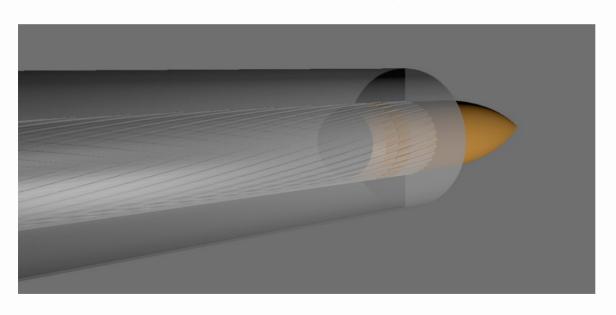


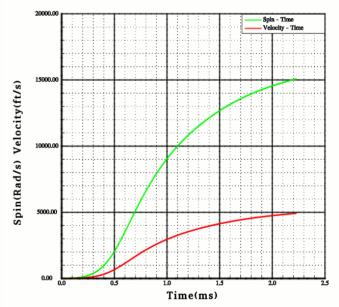
Our Procedure

Stage-1: Preparation of Draft FCI & TFT

Internal Ballistic Calculations

· Determination of standard muzzle velocities for each charge using CAEeda software.

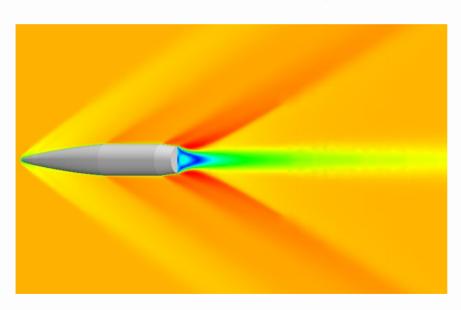


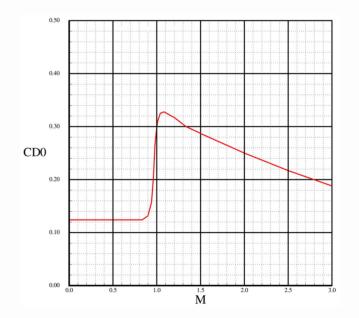


Stage-1: Preparation of Draft FCI & TFT

Aerodynamics calculations

· Computations of aerodynamic coefficients of projectile using CAEeda software

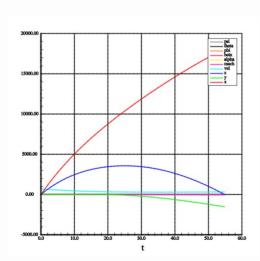


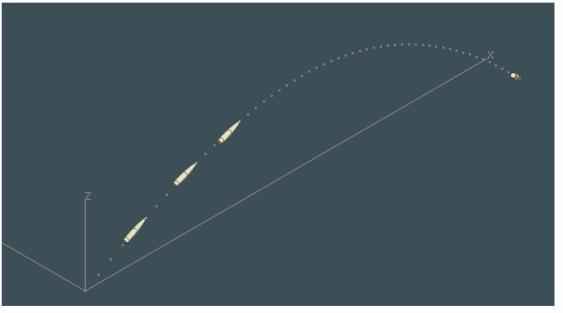


Stage-1: Preparation of Draft FCI & TFT

Flight mechanics calculations

- · Modified Point Mass Simulations
- · Calculations of max and min ranges and QE limits and vertex
- · Test safety calculations

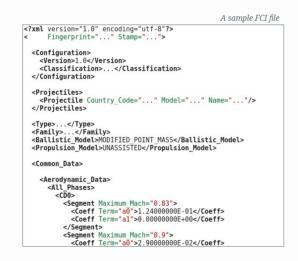




Stage-1: Preparation of Draft FCI & TFT

Preparation of draft FCI files and TFT with respect to NATO STANDARDS

FCI files and TFTs are created by recursive and interactive use (automation) of computer code developed by EDA for this purpose specifically.



								A sample
		BASIC	DATA AND	TABLE F (i) CORRECTI		O BEARIN	G	
J, HE, E, PD,								CHARG V ₀ = 317.2
1	2	3	4	5	6	7	8	9
RANGE	QUADRANT ELEVATION	FUZE SETTING	CORRECTION TO FUZE	EFFECT ON RANGE	FORK	TIME OF FLIGHT	CORRECTIONS (Δ _C A	
	FOR STANDARD CONDITIONS	FOR GRAZE BURST	SETTING TO CHANGE HEIGHT OF BURST DOWN BY 10M	FOR INCREASE OF ONE MIL IN ELEVATION			DRIFT (CORRECTION TO LEFT)	1 KNOT CROSSWIND
(X)	(A _{QE})	(FS)	(Δ _C FS/ -10M Y _b)	(ΔX/ 1 MIL A _{QE})	(F)	(TOF)	(A_d)	(1KT W _z)
М	MIL			М	MIL	S	MIL	MIL
0	0.0				0	0.0	0.0	0.00
100	5.1	0.3	1.24		0	0.3	0.1	0.00
200	10.0	0.7	1.10		0	0.7	0.2	0.00
300	15.0	1.1	0.97		0	1.1	0.3	0.00
400	20.1	1.3	0.93	20	0	1.3	0.2	0.02
500	25.2	1.6	0.84	19	0	1.6	0.3	0.02
600	30.4	1.9	0.77	19	1	1.9	0.3	0.03
700	35.6	2.3	0.70	19	1	2.3	0.4	0.03

Firing Tests for Data Acquisition (FTDA)



Stage-2: Development of FCI & TFT

Firing test data analysis

- · Data organization
- · Data reduction by specific statistical methods

Determination of standard muzzle velocities from firing test data

· Performed for all charges

Finding muzzle velocity correction with respect to projectile mass

· Performed for all charges



Stage-2: Development of FCI & TFT

Finding correction factors for aerodynamic coefficients with respect to quadrant elevations

· Performed for all charges

Determining probable errors from firing test data using the error budget analysis

- · Calculation PE in range and deflection
- · Calculation of fork (for up-leg, vertex, down-leg of trajectory)
- · Performed for all charges

FCI files are created by recursive and interactive use (automation) of computer code developed by EDA for this purpose specifically.

Stage-2: Development of FCI & TFT

Preparing TFTs and FCI files for projectile, weapon and fuze with respect to NATO STANDARDS

TFTs are created by recursive and interactive use (automation) of computer code, which accepts FCI files as input, developed by EDA for this purpose specifically.

```
<?xml version="1.0" encoding="utf-8"?>
     Fingerprint="..." Stamp="...">
 <Configuration>
   <Version>1.0</Version>
   <Classification>...</Classification>
 </Configuration>
 <Projectiles>
   <Projectile Country_Code="..." Model="..." Name="..."/>
 </Projectiles>
 <Type>...</Type>
 <Family>...</Family>
 <Ballistic Model>MODIFIED POINT MASS</Ballistic Model>
 <Propulsion Model>UNASSISTED</Propulsion Model>
 <Common Data>
   <Aerodynamic Data>
     <All Phases>
          <Segment Maximum Mach="0.83">
           <Coeff Term="a0">1.24000000E-01</Coeff>
           <Coeff Term="a1">0.00000000E+00</Coeff>
          <Segment Maximum Mach="0.9">
            <Coeff Term="a0">2.90000000E-02</Coeff>
```

		BASIC	T DATA AND	ABLE F (i) CORRECTI		O BEARIN	G		
I, HE, , PD,								CHARG V ₀ = 317.2	
1	2	3	4	5	6	7	8	9	
RANGE	QUADRANT ELEVATION FOR STANDARD CONDITIONS	FUZE SETTING FOR GRAZE BURST	CORRECTION TO FUZE SETTING TO CHANGE HEIGHT OF BURST DOWN BY 10M	EFFECT ON RANGE FOR INCREASE OF ONE MIL IN ELEVATION	FORK	TIME OF FLIGHT	CORRECTIONS (AcA) DRIFT (CORRECTION TO LEFT)		
(X)	(A _{QE})	(FS)	(Δ _C FS/ -10M Y _b)	(ΔX/ 1 MIL A _{QE})	(F)	(TOF)	(A _d)	(1KT W _z)	
М	MIL			М	MIL	S	MIL	MIL	
0	0.0				0	0.0	0.0	0.00	
100	5.1	0.3	1.24		0	0.3	0.1	0.00	
200	10.0	0.7	1.10		0	0.7	0.2	0.00	
300	15.0	1.1	0.97		0	1.1	0.3	0.00	
400	20.1	1.3	0.93	20	0	1.3	0.2	0.02	
500	25.2	1.6	0.84	19	0	1.6	0.3	0.02	
600	30.4	1.9	0.77	19	1	1.9	0.3	0.03	
700	35.6	2.3	0.70	19	1	2.3	0.4	0.03	

A sample FCI file

A sample TFT

Firing Tests for Verification and Acceptance (FTVA)



About FCI Files

FCI Files

- > Needed for fire control computer
- > Consisting of four different files
- > Contains ballistic performance information of the Gun & Bullet duo
- > Must be rebuilt if the weapon or ammunition changes
- > Required to create TFT

Weapon File

- Bore diameter, barrel length, riffle twist

Fuse File

- Type and specifications of the fuse

Projectile File

- Aerodynamic coefficients
- Diameter, mass and inertial moments
- "Fitting" inputs
- "Management" inputs
- "Probable Error" inputs
- Charge performance inputs
- Basebleed performance inputs
- Weapon information inputs
- "Fitting", "Management", "Probable Error" and weapon information inputs are calculated separately for each propellant charge

A Sample FCI File

```
*Processing Instruction* <?xml xml(version="1.0" encoding="utf-8")
▼-<NABK Fingerprint="..." Stamp="..."
    ▼-1 <Configuration
          -1 < Version 1.0
          -2 < Classification .../...
    ▼-2 < Projectiles
         1<Projectile Country_Code="..." Model="..." Name="..."
      -3 <Type HEA
      4 <Family ...
     5 < Ballistic Model MODIFIED POINT MASS
     -6 < Propulsion_Model UNASSISTED
    ▼-7 <Common_Data
        ▼-1 <Aerodynamic_Data
            ▼-1 <All_Phases
▼-1 <CD0
                     ▼-1<Segment Maximum Mach="0.8.."
                            1 Coeff Term="a0" 124000000E-01
                           -2 < Coeff Term="a1" 0.00000000E+00
                  ▶-2 <CDA2
                  ▶-3 <CLA
                  ▶-4 <CLA3
                  5 < CMA
                 -6 < CMAGF
-7 < CSPIN
        ▼-2 <Fuze_Data
             ▶-1 <Fuze Country_Code="..." Model="..."
        ▼-3 <Projectile_Characteristics
               -1 < Diameter 0.1..
               -2 < Delta Square Weight 0.6..
               -3 <Axial_Moment_Of_Inertia 0.1..
               4 < Reference_Fuze_Mass 0.8..
               5 < Reference_Mass 4...
             ▶ -6 <Squares
       ▼-4 <Fitting_Data
▼-1 <All_Phases
▼-1 <Constants
                  2 < DELUM
                  -3 < DELUPT
                 -4 <FL
-5 <IFORMQE
-6 <TCORR
        → -5 < ROURN

→ -5 < MV_Management_Data

→ -1 < EROSION

→ 2 < MVLOSS

→ 3 < MVN

→ 4 < PEOCC
        ▼-6 <Probable_Error_Data
            -1 <FORK
-2 <PED
-3 <PEI
              4 < PEM 0.1..
             ▶ -5 < PEMV
             6 < PEQE
    ▼-8 <Weapon_Specific_Data
        ▼-1 <Weapon Type="HOWITZER" Model="..."
             ▶ -1 < Charge_Performance_Data
             -2 <Safety_Data
```

About TFT

These are the tables used to calculate the elevation and bearing of the barrel in order to hit the target, and fuse settings.

They consist of A, B (B1, B2), C, D, E, F (F1, F2, F3), G (G1, G2), H, I, J(J1, J2), K tables.

TFT Samples

11100	arreptes									
			BASIC	DATA AND	TABLE F (i)		O BEARIN	G		
PROJ FUZE									CHARG V ₀ = 317.2	
	1	2	3	4	5	6	7	8	9	1
	RANGE	QUADRANT ELEVATION	FUZE SETTING	CORRECTION TO FUZE	EFFECT ON RANGE	FORK	TIME OF FLIGHT	CORRECTIONS (Δ _C A		
		FOR STANDARD CONDITIONS	FOR GRAZE BURST	SETTING TO CHANGE HEIGHT OF BURST DOWN BY 10M	FOR INCREASE OF ONE MIL IN ELEVATION			DRIFT (CORRECTION TO LEFT)	1 KNOT CROSSWIND	
	(X)	(A _{QE})	(FS)	(Δ _C FS/ -10M Y _b)	(ΔX/ 1 MIL A _{QE})	(F)	(TOF)	(A_d)	(1KT W _z)	
	М	MIL			М	MIL	s	MIL	MIL	1
	0	0.0				0	0.0	0.0	0.00	1
	100	5.1	0.3	1.24		0	0.3	0.1	0.00	1
	200	10.0	0.7	1.10		0	0.7	0.2	0.00	
	300	15.0	1.1	0.97		0	1.1	0.3	0.00	
	400	20.1	1.3	0.93	20	0	1.3	0.2	0.02]
	500	25.2	1.6	0.84	19	0	1.6	0.3	0.02	1

J, HE, E, PD,	CORF	RECTIO	NS TO R		BLE F (ii) FOR NON		OARD CO	NDITIO		CHAR(₀ = 317.2	
1	10	11	12	13	14	15	16	17	18	19	l
RANGE				RA	NGE COF	RECTI	ONS	(Δ _c X)			ĺ
	MUZ VELO		RANGE WIND		BALLISTIC AIR TEMPERATURE		BALLISTIC AIR DENSITY		PROJ MASS		
(X)	(V ₀)		(W _x)		(T _B)		(D _B)		(2 SQ STD)		
	(1 M/S) (1 KT)			(1%	6)	(1%)		(1 SQ)			
İ	DEC	INC	HEAD	TAIL	DEC	INC	DEC	INC	DEC	INC	ĺ
	(-)	(+)	(W)	(<u>W</u>)	(-)	(+)	(-)	(+)	(-)	(+)	
М	М	M	M	M	М	М	М	М	M	M	
0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0	0	
100	0.7	-0.7	0.1	0.0	0.1	0.0	0.0	0.0	-2	2	l
200	1.4	-1.5	0.1	0.0	0.2	0.0	0.0	0.0	-3	3	l
300	2.1	-2.2	0.1	0.0	0.3	-0.1	-0.1	0.1	-4	4	

Tabular Fire Tables (TFT)

- > Prepared by using FCI
- > Prepared for a weapon and bullet combination specifically
- > Based on standard atmospheric conditions (ICAO)
- > Prepared in a specific format in NATO standards
- > Prepared different approaches for direct and indirect firing and each propellant charge separately
- > Calculations for aiming can be done manually by using TTF, but it requires experience, and non-computerized manual calculations take time.
- > TFT-based calculations are not as accurate as FCI-based computational simulations (nonlinear), as they require a lot of interpolation (a linear approach).

Thank you



EDA Engineering Design & Analysis