

Our Ballistic Capabilities

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Ballistic Design and Analysis Topics

Ammunition design, optimization and analysis

Heavy weapon ammunitions

Small weapon ammunitions

Rocket

Weapon system/subsystem design, optimization and analysis

Howitzer

Tank

Mortar



Ballistic Capabilities

Internal Ballistics

External Ballistics

Terminal Ballistics

Projectile/Barrel Design & Design Optimization

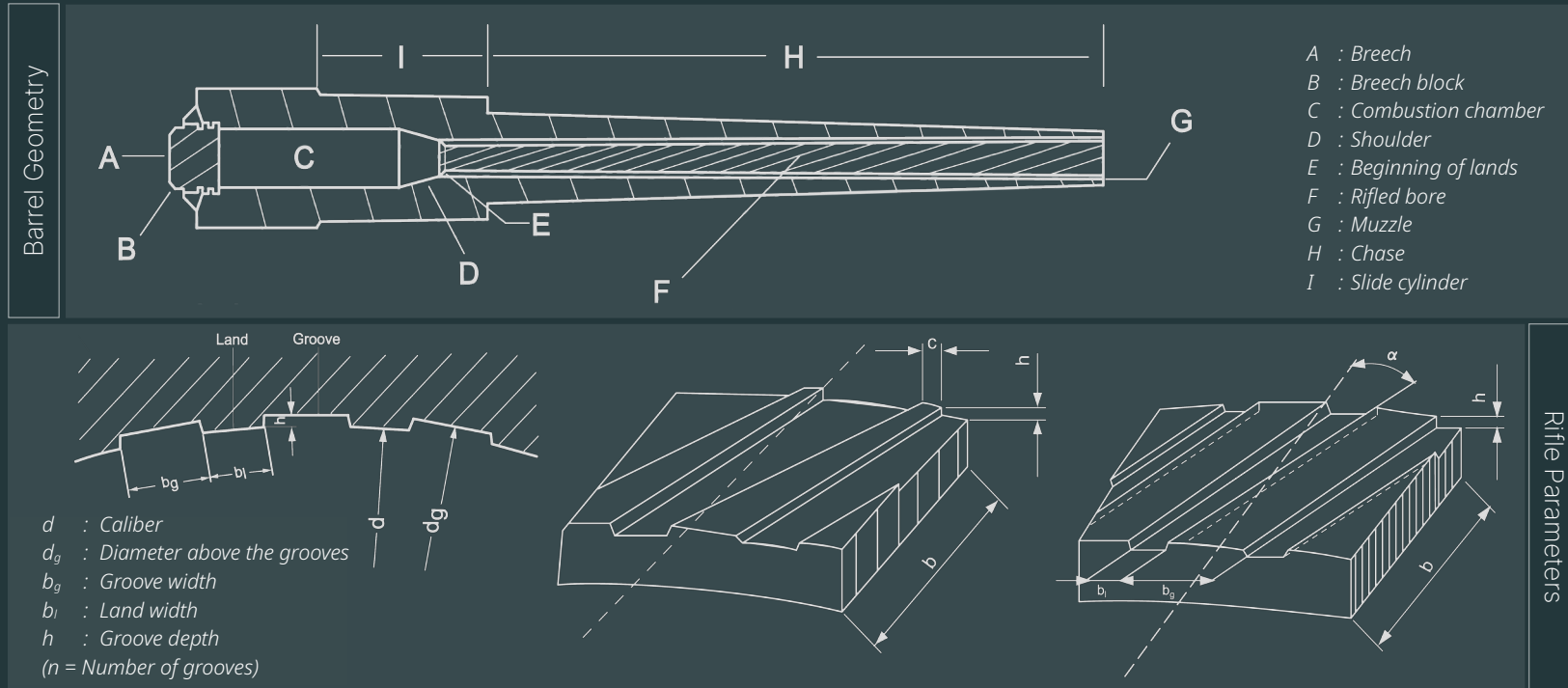
All studies are performed using BALLISTICeda and CAEeda

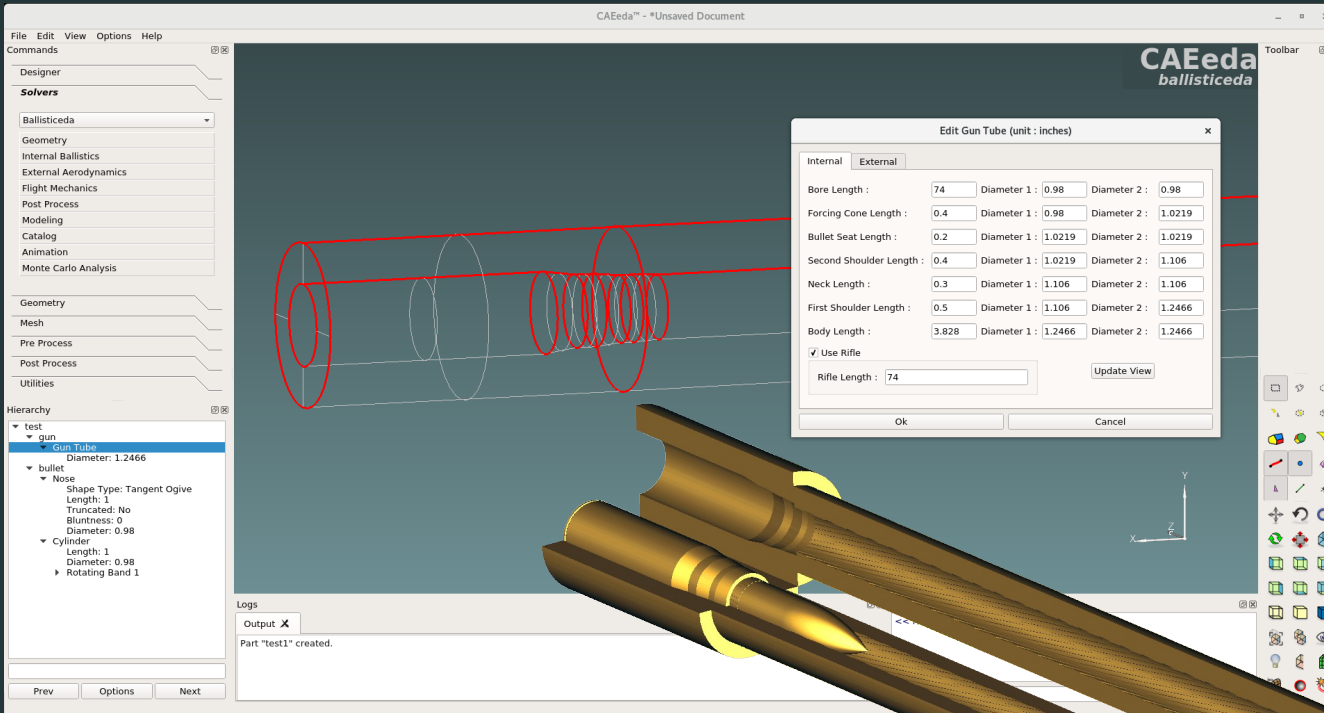
Internal Ballistics

- › Computation of propellant burning
- › Computation and simulation of projectile motion in a rifled gun bore
- › Structural and thermal analysis of gun barrels
- › Computational analysis of base-bleed



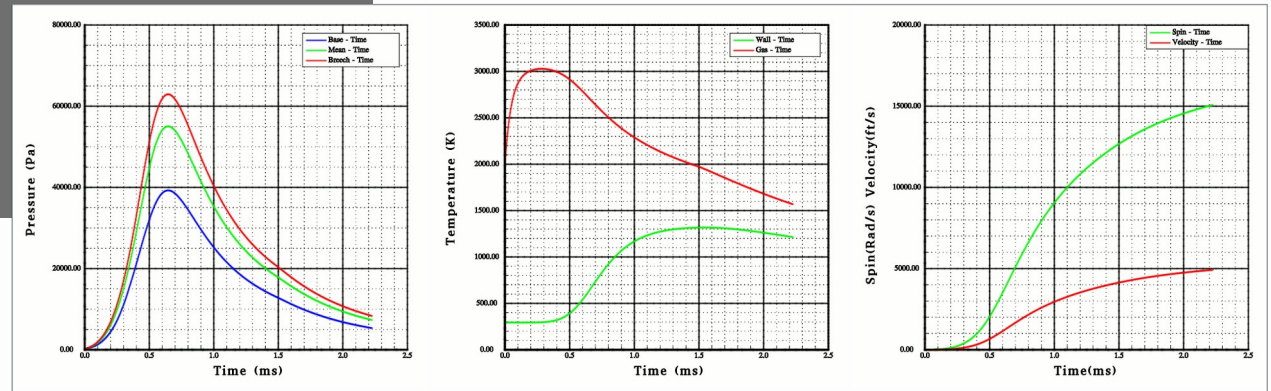
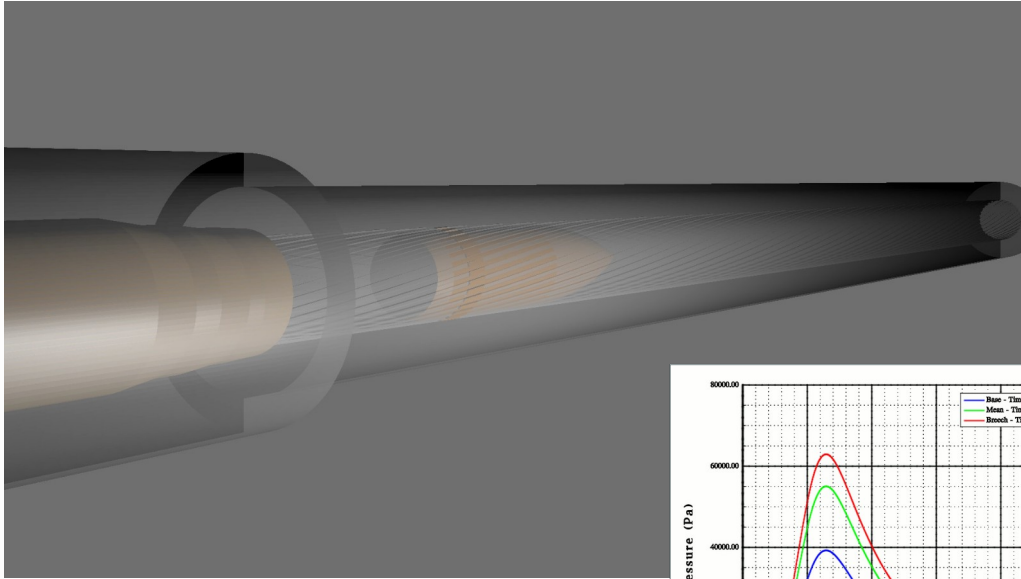
An example of an internal ballistics application



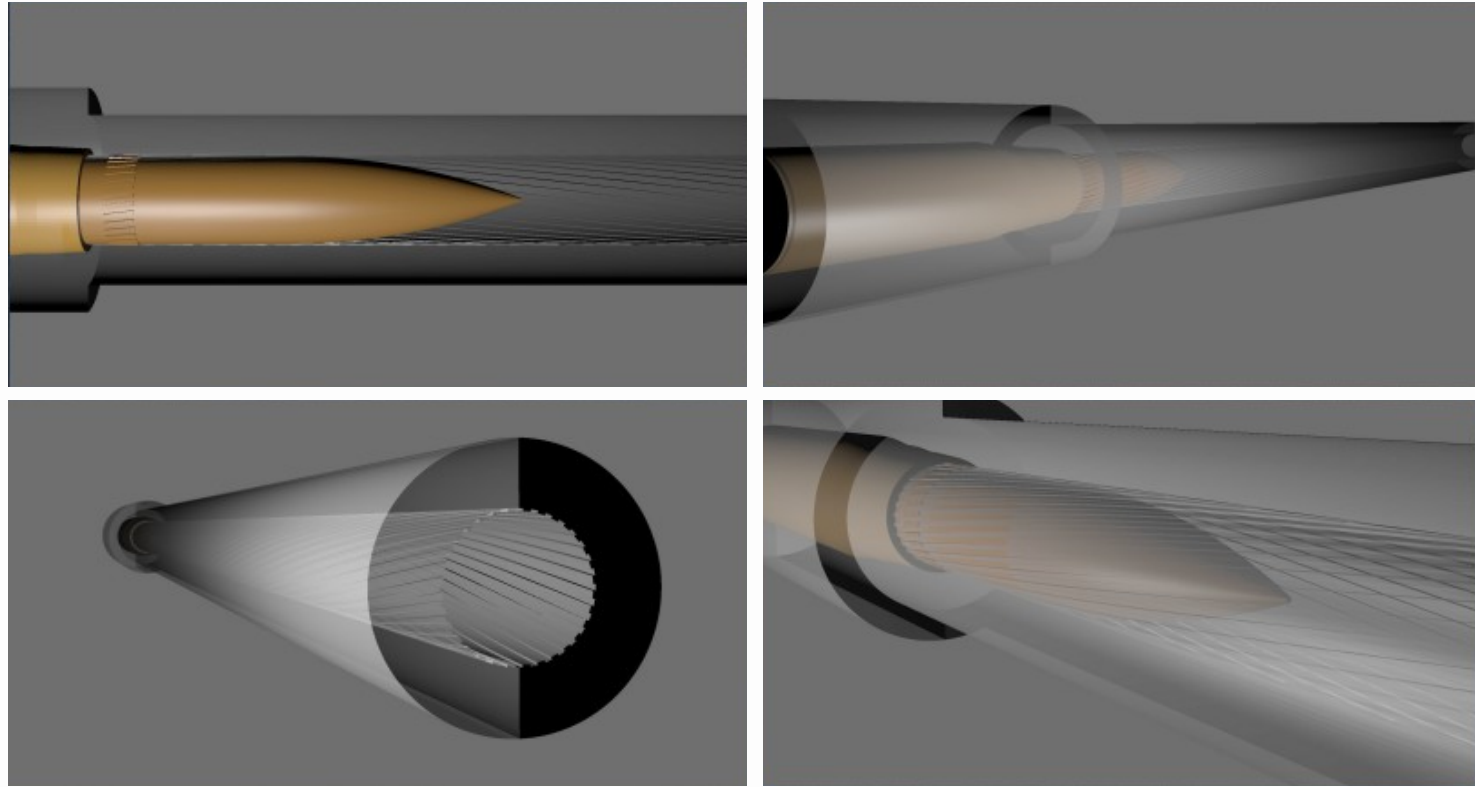


Parametric barrel design using BALLISTICeda

The simulation of projectile motion in a rifled tube

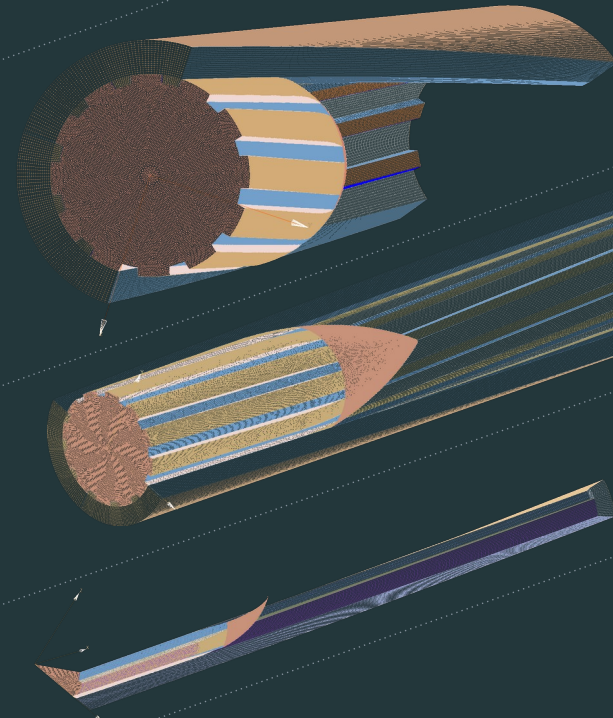
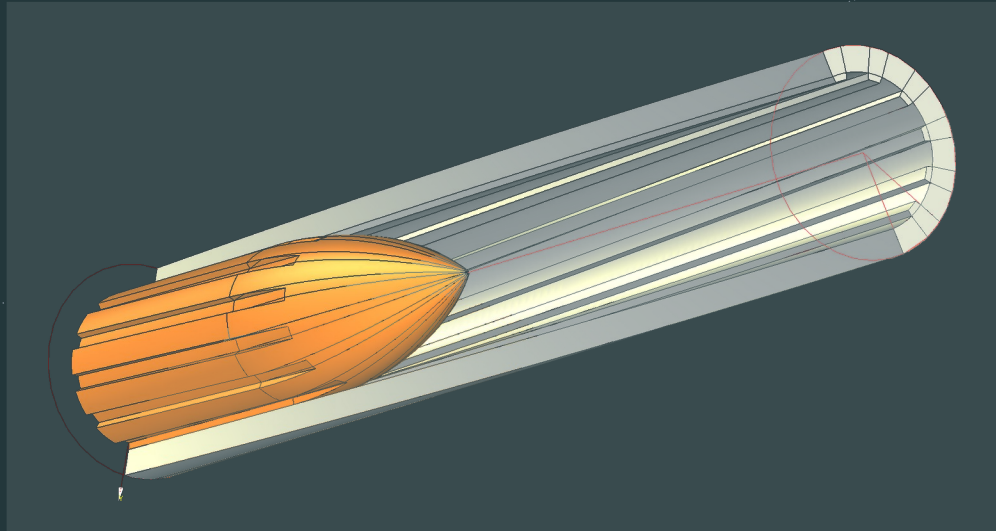


The simulation of projectile motion in a rifled tube



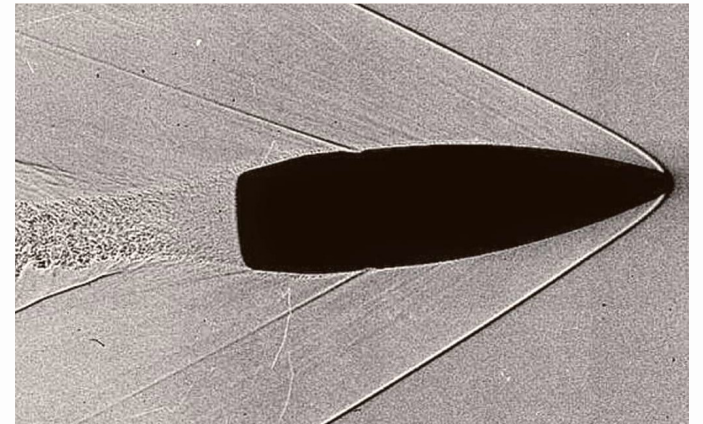
Internal Ballistics - Structural Mechanics Interaction

Method : Dynamic mesh coupling

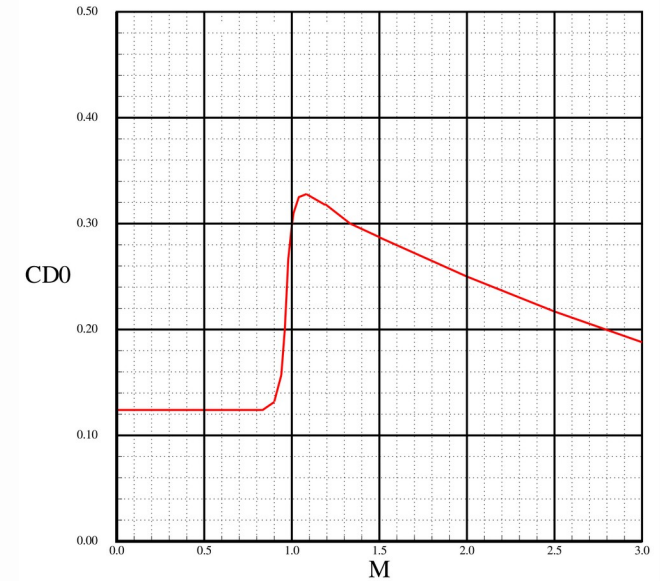
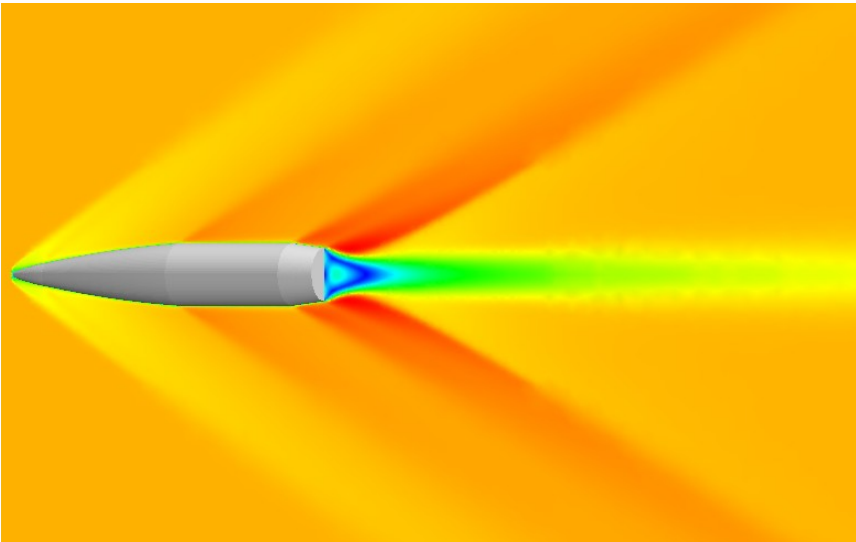


External Ballistics

- › Calculation of aerodynamic coefficients (static and dynamic) of projectiles and missiles using engineering methods and CFD
- › Flight simulation of spinning missiles and projectiles by using aerodynamic tables or CFD coupling
- › Extraction of aerodynamic coefficients from radar data



Calculation of aerodynamic coefficients using FAPeda (CFD module of CAEeda)



Calculation of aerodynamic coefficients using engineering methods (BALLISTICeda)

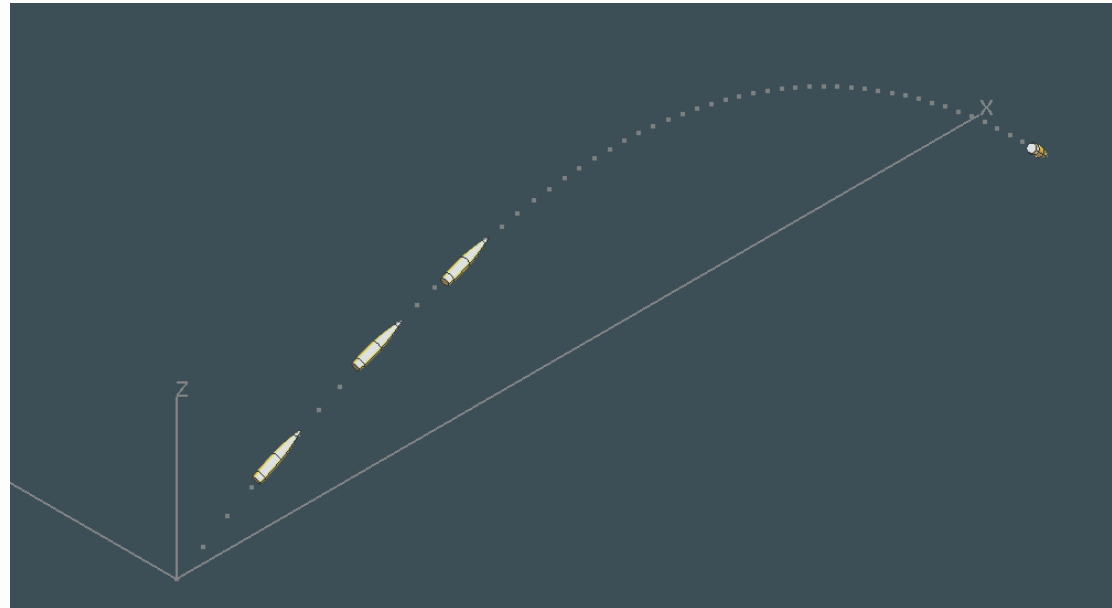
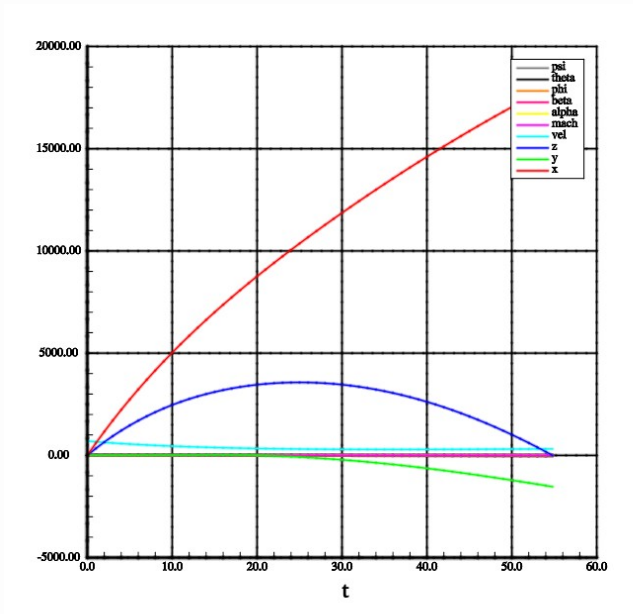
The screenshot displays the CAEeda software interface for a missile design. The main window shows a 3D model of a yellow missile with four fins. A dialog box titled "Define Aerodynamic Data" is open, showing the "Coefficients" tab. The "Body-Wind Axis System" section has several coefficients checked, including C_x (Axial force coefficient), C_y (Side force coefficient), C_z (Normal force coefficient), C_m (Roll moment coefficient), C_n (Pitch moment coefficient), C_p (Yaw moment coefficient), X_{cp} (Center of Pressure), C_D (Drag coefficient), and C_L (Lift coefficient). The "Derivatives" section has several checked, including $C_{n\alpha}$ (Derivative of Normal force coefficient wrt α), C_{yp} (Derivative of Side force coefficient wrt β), $C_{m\alpha}$ (Derivative of Pitch moment coefficient wrt α), $C_{n\beta}$ (Derivative of Roll moment coefficient wrt β), $C_{m\beta}$ (Derivative of Pitch moment coefficient wrt β), $C_{m\dot{\alpha}}$ (Derivative of Pitch damping coefficient wrt $\dot{\alpha}$), $C_{m\dot{\beta}}$ (Derivative of Pitch damping coefficient wrt $\dot{\beta}$), $C_{m\dot{\gamma}}$ (Derivative of Pitch damping coefficient wrt $\dot{\gamma}$), $C_{m\dot{\delta}}$ (Derivative of Pitch damping coefficient wrt $\dot{\delta}$), $C_{m\dot{\epsilon}}$ (Derivative of Pitch damping coefficient wrt $\dot{\epsilon}$), $C_{m\dot{\zeta}}$ (Derivative of Pitch damping coefficient wrt $\dot{\zeta}$), $C_{m\dot{\eta}}$ (Derivative of Pitch damping coefficient wrt $\dot{\eta}$), $C_{m\dot{\theta}}$ (Derivative of Pitch damping coefficient wrt $\dot{\theta}$), $C_{m\dot{\phi}}$ (Derivative of Pitch damping coefficient wrt $\dot{\phi}$), $C_{m\dot{\chi}}$ (Derivative of Pitch damping coefficient wrt $\dot{\chi}$), $C_{m\dot{\psi}}$ (Derivative of Pitch damping coefficient wrt $\dot{\psi}$), $C_{m\dot{\omega}}$ (Derivative of Pitch damping coefficient wrt $\dot{\omega}$), $C_{m\dot{\nu}}$ (Derivative of Pitch damping coefficient wrt $\dot{\nu}$), $C_{m\dot{\xi}}$ (Derivative of Pitch damping coefficient wrt $\dot{\xi}$), $C_{m\dot{\eta}}$ (Derivative of Pitch damping coefficient wrt $\dot{\eta}$), $C_{m\dot{\theta}}$ (Derivative of Pitch damping coefficient wrt $\dot{\theta}$), $C_{m\dot{\phi}}$ (Derivative of Pitch damping coefficient wrt $\dot{\phi}$), $C_{m\dot{\chi}}$ (Derivative of Pitch damping coefficient wrt $\dot{\chi}$), $C_{m\dot{\psi}}$ (Derivative of Pitch damping coefficient wrt $\dot{\psi}$), $C_{m\dot{\omega}}$ (Derivative of Pitch damping coefficient wrt $\dot{\omega}$), $C_{m\dot{\nu}}$ (Derivative of Pitch damping coefficient wrt $\dot{\nu}$), $C_{m\dot{\xi}}$ (Derivative of Pitch damping coefficient wrt $\dot{\xi}$).

A graph in the top right corner shows the aerodynamic coefficients versus Mach number (MACH) for an angle of attack $\alpha = 1$ deg. The graph plots C_x , C_y , C_z , C_m , C_n , C_p , C_D , and C_L against MACH from 0 to 1.5. The y-axis ranges from -400 to 400. The graph shows that C_x and C_D are positive, while C_y , C_z , C_m , C_n , C_p , and C_L are negative. The coefficients generally decrease in magnitude as Mach number increases.

The left sidebar shows the "Hierarchy" tree with the following structure:

- Geometry1
 - missile
 - Nose
 - Shape Type: Tangent Ogive
 - Length: 203.21
 - Truncated: No
 - Bluntness: 0
 - Diameter: 66.04
 - Material: Steel_ST36
 - Cylinder
 - Length: 787.39
 - Diameter: 66.04
 - Material: Steel_ST36
 - Finset 1
 - Body Offset: 695
 - Radial Offset: 0
 - Material: Steel_ST36
 - Fin 1
 - Base
 - Profile: Wedge
 - Base Length: 93.34
 - Base Angle: 0
 - Section 1
 - Profile: Wedge
 - Chord Length: 57.2
 - Span Length: 86.6
 - Sweep Angle: 21.5
 - Tilt Angle: 0
 - Twist Angle: 0
 - Fin 2
 - Fin 3
 - Fin 4
 - Finset 2

Flight mechanics simulations using BALLISTICeda



Terminal Ballistics

- › Dispersion analysis
- › Hit target analysis



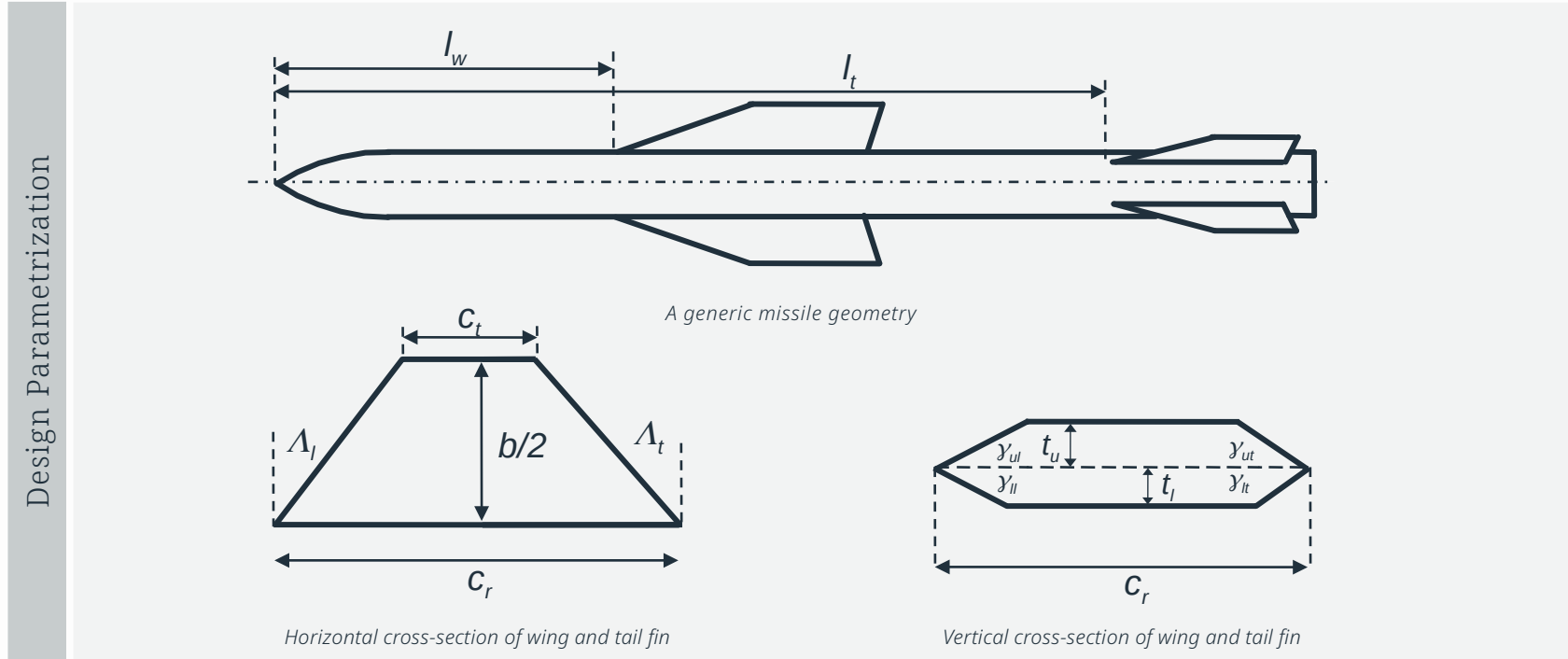
Projectile and Barrel Design & Design Optimization

- › Aerodynamic shape optimization of projectiles
- › **Design and optimization** of kinetic energy projectiles
- › Sabot design
- › Structural/thermal design and optimization of barrels

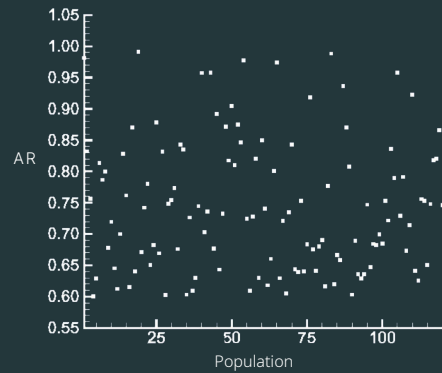


Example : Airframe shape optimization for range maximization

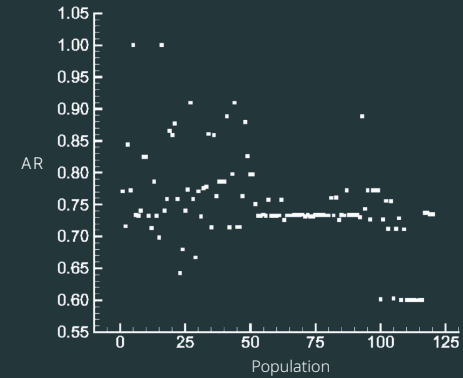
Missile Design Optimization Using Genetic Algorithm



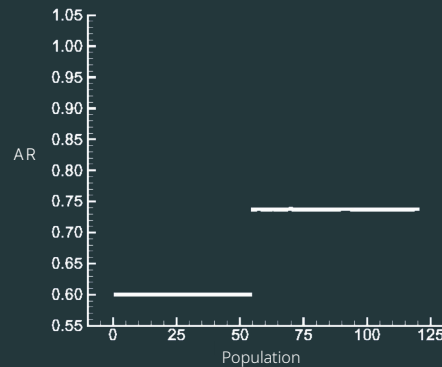
Design Steps



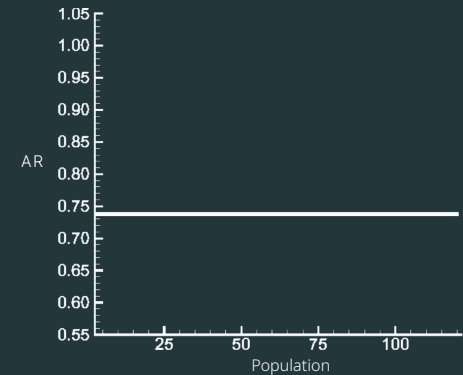
(1) Initial aspect ratio of the missile tail fin



(2) Intermediate aspect ratio of the missile tail fin



(3) Intermediate aspect ratio of the missile tail fin

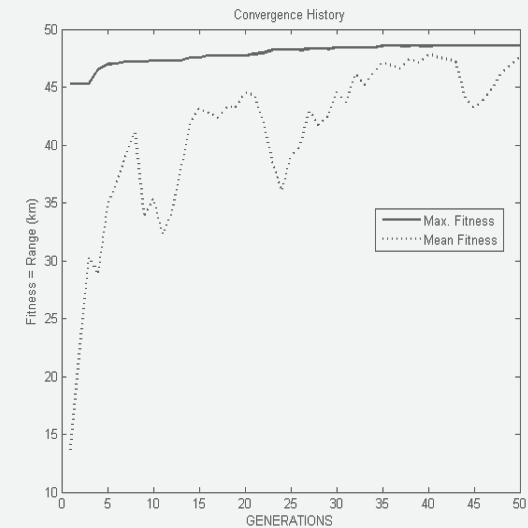
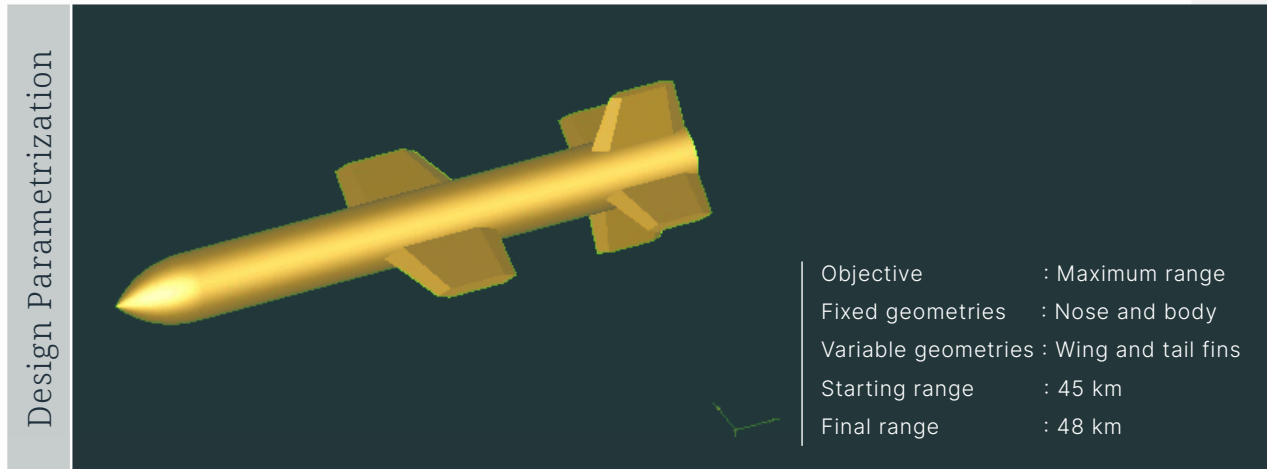


(4) Optimum aspect ratio of the missile tail fin

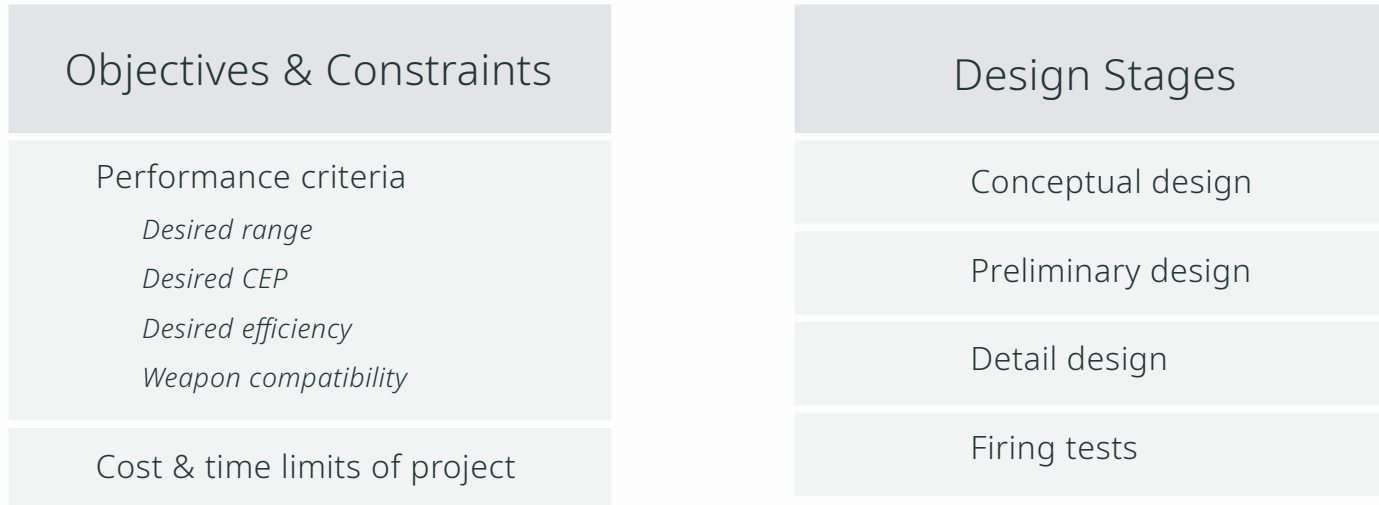
CAD, MESH, CFD Coupling for Design Automation and Optimization

Sample Application : Wing and Tail Shape Optimization to Extend the Range of a Missile

Final Design



Our Ballistic Design Approach



The Scope of Ballistic Design & Analysis

Geometric modeling

Aerodynamic analysis

Internal ballistic analysis

Flight mechanics analysis

Structural mechanics analysis

Thermal analysis

Optimization

Design automation

Post processing (plots & animations)



Tools and Methods Used

- › Geometric modeling
 - *Automatic and parametric 2D & 3D CAD modeling using the Geometry Module of CAEeda*
 - *Automatic and parametric projectile & missile modeling using the Geometry Module of BALLISTICeda*
- › Aerodynamic analysis
 - *Rough engineering solutions using the Aerodynamics module of BALLISTICeda*
 - *Precise CFD solutions using FAPeda module of CAEeda code*
- › Internal Ballistic Analysis
 - *Rough engineering solutions using the Intenal-Ballistics module of BALLISTICeda*
 - *Precise CFD solutions using the FAPeda module of CAEeda*
- › Flight Mechanics Analysis
 - *3DOF, Modified Point Mass, 5DOF, 6DOF solutions using Flight Mechanics module of BALLISTICeda*
- › Structural mechanics analysis using the SAPeda module of CAEeda
- › Heat transfer analysis using the TAPeda module of CAEeda
- › Design optimization by designer module of CAEeda
- › Post processing using POSTeda module of CAEeda

Thank you



EDA Engineering Design & Analysis